

# Miniflo-L™

## Digital Fuel Management System

RS-232 output format  
with interface to LORAN-C and GPS receivers



## **OPERATING MANUAL** Single and Twin Engine Indicators

For P/N: 91204XT-D





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NOTE: Though references are made in this manual to fuel measured in gallons, the information applies equally to measurements in pounds, kilos, or liters.

# **Miniflo-L™**

Although not required by the FAA, it is recommended that this manual be attached to the FAA-approved Flight Manual, or always kept on board for reference.

## **1. GENERAL DESCRIPTION**

Miniflo-L™ is a Digital Fuel Management System designed to provide complete fuel management information under real flight conditions without any manual entry of data (after entry of the initial fuel on board information).

Miniflo-L™ is connected to the engine fuel flow transducer for fuel flow information and to the Loran-C or GPS receiver serial port for navigation data (ground speed, distance and estimated time en route).

This system is also capable of transmitting the fuel information to the Bendix/King KLN-88, KLN-90, and Garmin GPS navigation receivers, for additional calculations and display of fuel management data.

Miniflo-L™ is set up to measure the flow of fuel in either gallons, liters, pounds, or kilos, and it can be installed on virtually any reciprocating or turbine engine by selecting the proper size fuel flow transducer.

## **1.1 THE SYSTEM PROVIDES**

### **1.1.1 SPECIFIC RANGE**

Specific Range (NM/GAL) = Ground Speed (Kt.)/Fuel Flow per hour. This is an indication of how efficient the cruise is and the optimum cruise speed can be determined by selecting the power setting, which yields the highest nautical miles per gallon.

### **1.1.2 FUEL TO DESTINATION**

Miniflo-L™ calculates (under real wind conditions) the fuel necessary to reach the destination as selected on the Loran-C or GPS receiver by multiplying the fuel flow by the estimated time en route to the destination. (If an intermediate waypoint is selected for navigation purposes, the displayed fuel to destination represents the fuel needed to reach the intermediate waypoint unless the distance to the destination is in the serial message.)

### **1.1.3 FUEL RESERVE**

Miniflo-L™ calculates the amount of fuel, which will be available on board when the aircraft reaches the destination or waypoint indicated on the Loran-C or GPS receiver. This feature provides the pilot with accurate information so that the reserve fuel situation can be evaluated and action can be taken if necessary. (If an intermediate waypoint is selected for navigation purposes, the displayed fuel reserve represents the fuel reserve at the intermediate waypoint unless the distance to the destination is in the serial message.) Fuel Reserve = Fuel Remaining – Fuel to Reach Destination.

### **1.1.4 ENDURANCE**

Miniflo-L™ calculates the time left to fly in hours and minutes based on the fuel remaining and the present fuel flow.

#### 1.1.5 FUEL FLOW

The system provides a digital readout of the fuel per hour to a tenth of a gallon up to 100 gallons and to the nearest gallon above 100 gallons. For the pounds version, the readout is to the nearest pound up to 999 lbs./hour and to the nearest 10 lbs. above 999 lbs./hour.

#### 1.1.6 FUEL USED

The system keeps track of the fuel used since the last fuel entry or reset.

#### 1.1.7 FUEL REMAINING

The system keeps track of the fuel remaining on board. Fuel Remaining is equal to Initial Starting Fuel minus Fuel Used.



## **1.2 SYSTEM COMPONENTS**

The system consists of three (3) basic units: the fuel flow transducer, the navigation receiver (Loran-C or GPS) and the panel mounted unit.

### **1.2.1 FUEL FLOW TRANSDUCER**

The fuel flow transducer mounted in the fuel line measures the flow of fuel and generates electrical pulses directly proportional to the fuel flow. The transducer is fail-safe designed, and stopped rotor will not interrupt fuel flow to the engine.

### **1.2.2 LORAN-C OR GPS RECEIVER**

The Loran-C or GPS receiver provides ground speed, distance, and estimated time en route through the serial port.

### **1.2.3 PANEL MOUNTED UNIT**

All system electronics, function controls, and digital displays are contained in a single instrument that mounts in a standard 3¼" wide and 1¼" high opening. This unit requires no periodic maintenance, adjustment, or calibration once it is properly installed.

**The Display:** The fuel flow is always displayed on the left side of the display window. All other functions, with the priority for the rotary switch functions, are displayed on the right side of the display window.

**System Memory:** The system includes a non-volatile memory that retains fuel remaining and fuel used information when the power to the unit is shut down.

### **1.3 TEST FUNCTION**

Diagnostic software is built into the system. To activate it, press the ENTER/TEST button. All of the digits will be sequentially on in a rotating pattern for ten seconds. If the test is successful, the word “Good” will appear in the display window for three seconds. If the test is not successful, the word “bAd” and an error message identifying the error will be displayed. In such case, the unit will cease to function and must be considered unserviceable until corrective action is taken.

At the end of the test routine the system will display the following:

1. The K-factor for the flow transducer in the left window, and the display units (i.e. gal, P5.8, P6.7) in the right window. Note: In the twin-engine models the K-factor for the right engine appears in the right window of the next display screen.
2. Lt ( ): Loran or GPS type appears in the left window (without signal, will display “LOF”).
3. The distance to waypoint or destination is displayed as shown on the Loran-C or GPS receiver to check the Data Interface Integrity (not available without signal).
4. “FUL” appears in the left window and the maximum usable fuel in the right window.
5. Software version.
6. If the system cannot read the navigation receiver data, “Lbd” will display in the right window.

Note: Using the test function while engines are running will cause the computer to lose 17 seconds of fuel count.

## **2. PREFLIGHT PROCEDURES**

Miniflo-L™ is a fuel flow measuring system and NOT a quantity-sensing device. A visual inspection and positive determination of the usable fuel in the fuel tanks is a necessity. Therefore, it is imperative that the determined available usable fuel be manually entered into the system.

### **2.1 INITIAL PROGRAMMING**

The function of initial programming is to enter the total usable fuel into the memory. It can then be recalled whenever the fuel tanks are filled up to the maximum usable fuel. The “FUL” fuel setting determines the maximum amount of fuel that can be entered by any method into the Miniflo-L™.

#### **PROCEDURE:**

1. Power the unit up.
2. Move the FULL/ADD toggle switch to the FULL position and hold for the entire procedure.
3. Simultaneously press the ENTER/TEST button and move the REM/USED toggle switch to the REM position. The system will then countdown for 15 seconds, displaying the count on the left in the display window.
4. The word “FUL” and the current full fuel value will appear in the display window. Release the ENTER/TEST button and REM/USED toggle switch. Keep holding the FULL/ADD toggle switch in the FULL position.

5. Move the REM/USED toggle switch to the REM position to increment the full fuel number or to the USED position to decrement the number. (The longer you hold the switch in position, the faster the number will be updated.)
6. After reaching the correct total usable fuel figure, press the ENTER/TEST button and the computer will store that number as full fuel. The word “FUL” disappears and the computer will return to the operating mode. Release the FULL/ADD toggle switch.
7. To verify that the data is stored properly, press the ENTER/TEST button. The computer will run a diagnostic check and then display “Good”. If the test is successful, it will display the maximum usable fuel.

**NOTE: Do not turn the power off to the computer for approximately one minute. This will ensure that the unit has enough time to store the proper figures into the program.**

## **2.2 PREFLIGHT CHECK**

Initiate the diagnostic software built into the system by pressing the ENTER/TEST button; the program checks the hardware and the display. If the test is successful the word “Good” will appear in the display window; if not, the word “bAd” appears. The system is considered unserviceable until corrective action is taken.

At the end of the test routine, the system will display the following:

1. The K-factor for the flow transducer in the left window, and the display units (i.e. gal, P5.8, P6.7) in the right window. Note: In the twin-engine models the K-factor for the right engine appears in the right window of the next display screen.

2. Lt ( ): Loran or GPS type appears in the left window (without signal, will display “LOF”).
3. The distance to waypoint or destination is displayed as shown on the Loran-C or GPS receiver to check the Data Interface Integrity (not available without signal).
4. “FUL” appears in the left window and the maximum usable fuel in the right window.
5. Software version.
6. If the system cannot read the navigation receiver data, “Lbd” will display in the right window.

Note: Using the test function while engines are running will cause the computer to lose 17 seconds of fuel count.

Move the REM/USED toggle switch to the USED position. The system will display the fuel used since last fuel entry or fuel used since last reset.

Move the REM/USED toggle switch to the REM position. The system will display the fuel remaining on board. The pilot should confirm this figure with the actual fuel on board.

### **2.3 NO FUEL ADDED**

This automatically stores information concerning previous fuel levels, even in the case of a power down. If no fuel is added, no action is needed in updating fuel data.

## 2.4 FUEL TANKS FULL

There are two methods to enter full fuel: the ramping method and the FULL/ADD toggle switch method.

### **Ramping Method**

- Move the REM/USED toggle switch to the REM position and hold.
- Press the ENTER/TEST button to increment the fuel remaining until the total usable fuel is reached. (The longer you press, the faster the incrementing.)
- Release the REM/USED toggle switch and the ENTER/TEST button to enter the total usable fuel on board into memory.
- If the required figure is exceeded, follow the procedure in this manual, section 2.6 Correcting Fuel on Board Entry Error.

### **FULL/ADD Toggle Switch Method**

- Move the FULL/ADD toggle switch to the FULL position and hold.
- Press the ENTER/TEST button.
- Release the FULL/ADD toggle switch so it returns to the center position.
- To verify, move the REM/USED toggle switch to the REM position. Total usable fuel will be displayed.

## 2.5 PARTIAL FUEL ADDED

There are two methods to enter partial fuel:

### **Ramping Method**

Add the amount of fuel from the refueling meter to the amount of fuel remaining. Enter the total using the following steps:

- Move REM/USED toggle switch to REM position and hold.
- Press and hold ENTER/TEST button to increment fuel remaining until figure to be entered is reached; then release button.
- Release the REM/USED toggle switch. The displayed figure is entered into memory as fuel remaining on board.
- If the required figure is exceeded, follow the procedure in this manual, section 2.6 Correcting Fuel on Board Entry Error.

### **FULL/ADD Toggle Switch Method**

- Move FULL/ADD toggle switch to ADD position and hold.
- Move REM/USED toggle switch to REM position to increment fuel added figure until amount of fuel added is reached.
- Press the ENTER/TEST button.
- Release the FULL/ADD toggle switch so it returns to the center position. The computer will add the added fuel remaining and use the total as the current fuel remaining.
- To verify, move the REM/USED toggle switch to the REM position. The current usable fuel remaining will be displayed.

## **2.6 CORRECTING FUEL ON BOARD ENTRY ERROR**

In case an error has been made by exceeding the correct amount in entering the total usable fuel, move the REM/USED toggle switch to the USED position and simultaneously press and hold ENTER/TEST button. The fuel remaining figure will appear and pause in the display window for four (4) seconds. The figure will decrement (the longer you press, the faster it decrements). When the correct figure is reached, release both the REM/USED toggle switch and the ENTER/TEST button. To avoid repeating the four-second pause before decrementing, hold the REM/USED toggle switch in the USED position and use the ENTER/TEST button to control the decrementing.

Note: Adding or subtracting fuel by any method resets the fuel used value to zero.

## **3. INFLIGHT OPERATIONS**

### **3.1 INSTRUMENT OPERATION**

#### **3.1.1 FUEL FLOW**

For Single Engine Aircraft: FUEL FLOW is displayed continuously on the left display window.

For Twin Engine Aircraft: Total fuel flow is displayed continuously on the left side of the display window. To read each engine's fuel flow separately, move the FULL/ADD switch to FULL position, and the REM/USED switch to USED position, simultaneously. Left and right fuel flow will be displayed only as long as you hold the switches together and upon release the unit will return to combined fuel flow after 3 seconds.



### 3.1.2 FUEL USED

Fuel used is displayed by moving the REM/USED toggle switch to the USED position. The information is shown on the right display window as long as the switch is held in the USED position and for three seconds after the switch is released. The display represents the fuel used since last reset.

### 3.1.3 FUEL REMAINING

Fuel remaining is displayed by moving the REM/USED toggle switch to the REM position. The information is shown on the right display window as long as the switch is held in the REM position and for three seconds after the switch is released. The display represents the fuel remaining on board at the time of reading.

### 3.1.4 ENDURANCE

Endurance is selected by rotating the rotary switch to the ENDURANCE position. Endurance is displayed in hours and minutes on the right display window.

### 3.1.5 NAUTICAL MILES PER GALLON

Nautical miles per gallon is selected by rotating the rotary switch to the NM/GAL position. The information is shown on the right display window.

### 3.1.6 FUEL TO DESTINATION

Fuel to destination is selected by rotating the rotary switch to the FUEL TO DESTINATION position. The information is shown on the right display window and represents the fuel needed to reach either the active waypoint selected on the Loran-C (or GPS receiver) or the final destination (if the total distance record is provided in the serial message.) This assumes the aircraft ground speed and fuel flow remains constant and the aircraft remains on flight plan course. (Readings obtained during climb and descent are invalid.)

### **3.1.7 FUEL RESERVE**

Fuel reserve is selected by rotating the rotary switch to the FUEL RESERVE position. The information is shown on the right display window and represents the fuel that will be available when the aircraft reaches its destination as indicated on either the selected waypoint or the final destination (if the total distance record is provided in the serial message.) This assumes the aircraft ground speed, altitude, fuel flow, and direction remain constant. (Readings obtained during climb and descent are invalid.)

## **3.2 WARNINGS**

### **3.2.1 NOT ENOUGH FUEL**

When the rotary switch is in FUEL TO DESTINATION position, the information in the display window flashes if fuel on board is insufficient to reach either the destination selected as the active waypoint or the final destination (if that total distance record is provided in the serial message). Display window shows amount of fuel short to reach destination preceded by a negative sign.

### **3.2.2 RESERVE FUEL WILL BE USED**

With the rotary switch in the FUEL RESERVE position, the information in the display window flashes if the aircraft will arrive at the destination with less than the pre-programmed Endurance Warning Time minutes of fuel – calculated at the present cruise power setting. This warning is intended to alert the pilot that the prevailing condition will require the use of some of the selected reserve fuel to reach the destination.

### 3.2.3 LOW ENDURANCE

The Miniflo-L™ can be configured to display a warning based on the time remaining to fly. When the rotary switch is in the ENDURANCE position, and the actual endurance is less than the pre-programmed Endurance Warning Time, the data in the right half of the display flashes. Press the ENTER/TEST button to acknowledge the warning. (Note: Resetting the Miniflo-L™ or adding fuel resets this condition and the warning is enabled again).

### 3.2.4 LOW FUEL REMAINING

The system displays “Lo FUEL” when the fuel remaining reaches the pre-programmed Low Fuel Level configured in setup. Fuel flow information will not be displayed again until the pilot acknowledges this message by pressing the ENTER/TEST button. Fuel calculations are not interrupted by this message. (Note: Resetting the Miniflo-L™ or adding fuel resets this condition and the warning is enabled again).

## 4. EMERGENCY PROCEDURES

In case of electrical power failure in-flight, the instrument will cease to function. After restoring power, the system will resume accurate fuel flow reading, but time remaining, fuel used, fuel remaining, fuel reserve, fuel to destination and all warnings will not be accurate unless the duration of power failure is known and fuel consumption during the electric power failure is calculated and subtracted from fuel remaining.

## 5. ERROR MESSAGES

### ERROR 1:

Due to the necessity of *Group 1* settings, if the Flow Meter is set to Operate Mode and the checksum of *Group 1* is bad, the display will flash: **E1**.

This refers to Error 1, *Group 1*. The flow meter will not continue to function after this point, and will continue flashing **E1**, alerting that the flow meter must be serviced.

### ERROR 2:

The Flow Meter will still be accurate and operate under NON-LORAN pages, if the checksum of *Group 2* is bad. In this case, under any Loran page, the Loran information will be replaced by **E2**.

This display refers to Error 2, or *Group 2*. This is to alert the pilot that the Flow Meter does not have valid Loran and Output selections, and therefore, can not rely on Loran and Output information.

NOTE: Remember it is possible to set Group 2 settings without having to be in entry mode; therefore, this error can be fixed by going into *Manual Entry Mode*.

## 6. CONFIGURATION DATA ENTRY

### Manual Entry Mode

Ordinarily, the fuel flow indicator has been set up by the factory to match the K-factor of the supplied transducers and other set-up information. However, there are built-in provisions to change the set-up. Please be sure to define and document initial set-up before attempting to make changes.

#### Overview

Previously, all settings depended upon the switches mounted on the processor board. Currently, the Miniflo-L™ processor board and software version 60.01.XX has a feature that is referred to as *Manual Entry Mode*. In this mode, the Flow Meter settings are stored as two groups: *Group 1* and *Group 2* both shown in the table below.

Group 1	Group 2
Left & Right K Factors	Output Type (King, AirData, Arnav)
Fuel Units	Loran Input (On, or Off)
Single or Twin Engine Type	Endurance Warning Time (45, 30, 20, 10, or 5 minutes)
Low Flow Cutoff (On or Off)*	Filter Type (Injector or Carburetor)
Left & Right Fuel Flow Offset Frequency*	Low Fuel Level Warning (fuel level for warning to be issued)
	Ignore Loran Warning (Yes or No)

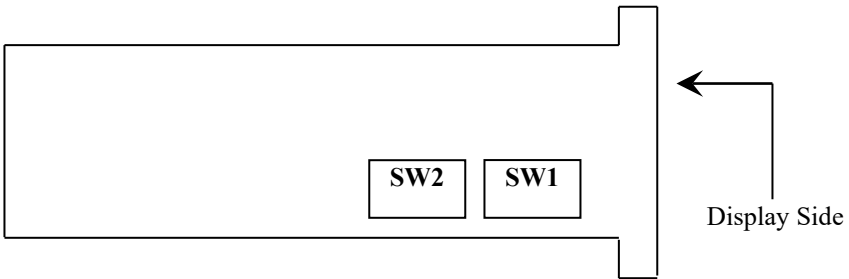
\*These functions are only applicable to DC systems.

Manual Entry Mode can be accessed in two ways: one providing access to both *Group 1* and *Group 2* values, and one providing access to only *Group 2* values. The access to *Group 2* values can be obtained while the unit is installed in the aircraft. Access to *Group 1*, however, requires removal of the unit to adjust switch settings.

Group 1: Generally, *Group 1* is set up by the distributor and contains information defined by the part number. However, although functions are defined here, do not change them without proper knowledge or they will affect performance of the indicator.

Group 2: *Group 2* must be set up by programming the unit in *Manual Entry Mode*. *Group 2* settings allow the user or installer to change Loran or GPS input and output parameters, endurance warning time, and fuel flow filtering types.

Locations of the switches for the Miniflo-L™ are as follows:



Each switch has 16 positions, 0-9, A, B, C, D, E, and F.

Note: A hole has been cut into the can to allow access to switches normally covered by the red K-factor sticker.

**Operation Mode vs. Entry Mode**

**FE:** If Switch 1 is set to F and Switch 2 is set to E, the unit is in *Entry Mode*. This is the only mode that will allow the setting of Group 1 values onto the non-volatile memory of the unit. In this mode, both groups can be set. Once installed in the aircraft, this mode is no longer accessible.

**FF:** Once the settings have been programmed, Switches 1 and 2 should be set to *FF*. This is the *Operation Mode*, which is required for normal operations. In this mode, settings previously recorded for Groups 1 and 2 will be utilized, and not the switches. Group 2 can still be accessed through the Manual Entry Mode, but Group 1 is not accessible.

Switch	Entry Mode	Operation Mode
1	F	F
2	E	F
3	0	0
4	0	0

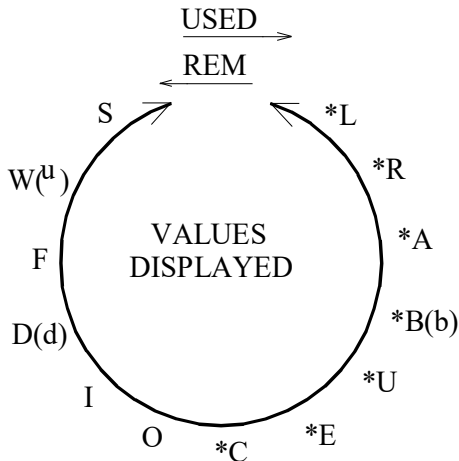
## Manual Entry Mode

There are two ways to access the Manual Entry Page.

1. Set Switches 1 and 2 to Entry Mode and power up. This allows access to both groups.
2. If the Switches are not set to Entry Mode, while running under normal conditions, press the ENTER/TEST button to start the test mode. When the version is displayed, press and hold the ENTER/TEST button for 15 seconds. This allows access to Group 2 only.

In both instances, “ENT” will be displayed. When “ENT” appears release button.

The display can now be paged through using the fuel USED button to scroll forward or the fuel REM button to scroll back.



**Note:** See page 25 for description of parameters.



The values displayed can be adjusted with the FULL/ADD toggle switch. ADD increments the value, and FULL decrements the value. As you hold ADD or FULL, the scrolling rate will increase up to a maximum speed.

If you wish to jump directly into the fastest scrolling speed, while holding the FULL/ADD toggle switch, press the fuel USED or fuel REM button.

Once the desired values are selected, press and hold the ENTER/TEST button while the upper window displays a countdown from 5 to 1. When the lower left window displays “SET,” release the ENTER/TEST button.

**Note:** It is recommended that you leave the unit powered up for at least one minute before turning the unit off. Reset switches 1 and 2 to Operate Mode (F,F) and reboot (Power ON). Then confirm the settings. The Manual Entry Pages will be displayed as follows. Symbols in ( ) represent 7 segment characters actually displayed.

### **Field K-factor adjust for Software Versions 60.01.72+**

1. Remove the Miniflo-L™ from the instrument panel.
2. Remove the red label from the top of the Miniflo-L™ and save it to be put back in place after the adjustment is complete.
3. With a small, flat blade screwdriver change the hexadecimal switch (SW2) closest to the rear connector from position “F” to position “E”.

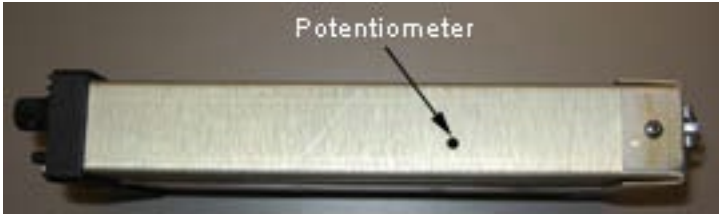
4. Reconnect the unit to the aircraft harness and turn the aircraft master switch on. An “L” for the left engine will appear on the left side of the left window and the most significant digit of the K-factor on the right side of the same window. The rest of the K-factor minus the least significant digit will appear in the right window.
5. The K-factor values displayed can be adjusted with the FULL/ADD toggle switch located at the lower right corner. ADD increments the value, and FULL decrements the value.
6. Move the REM/USED switch at the lower center position to the USED position to move to page “r” for the right engine in twin engine aircraft and repeat Step 5.
7. Press and hold the ENTER/TEST button (located in the lower left corner) position for a five seconds countdown from 5 to 1. When the word “SET” appears, release the button and wait 30 seconds before turning the aircraft master switch off.
8. Change the SW2 (hexadecimal switch) closest to the rear connector back to the “F” position. Put the red label back on the can as it was before the K-factor adjustment.
9. Turn the aircraft master switch on again to test the system for changes made. When the fuel flow settles down to zero, press and hold the ENTER/TEST button until the “8’s” start to move across the screens then release the button. The first screen after “Good” and “Shadin” will display the K-factor for the left engine in the left screen while the right screen shows the units of measure (i.e. gallons, pounds, etc). The next set of screens will be blank on the left while the right displays the K-factor for the right engine (if applicable). If the correct K-factor is displayed, the procedure was successful.

Display		Description
*L	xxxxx =	Left K-factor (where xxxxx is valid from 0 to 20,000. These are in 10s. A setting of 1234 would be a K-factor of 12,340)
*R	xxxxx =	Right K-factor (as above).
*A	xxxxx =	Left Fuel Flow Offset Frequency (Hz) for Analog Models Only
*B(b)	xxxxx =	Right Fuel Flow Offset Frequency (Hz) for Analog Models Only
*U	x =	Fuel Units are defined by the part number. Do not adjust these, as improper burn indication will occur. 0 = Gallons 1 = Liters 2 = Lbs 5.8 3 = Lbs 6.7 4 = Kilograms 5 = Lbs 6.5 6 = Lbs 6.35
*E	x =	Engine Type: 0 = Single 1 = Twin
*C	x =	Low Flow Cutoff: 0 = Off 1 = On  NOTE: Will not display fuel flow until a rate of 50 pounds per hour is reached.
O	x =	GPS/Output Type: Only used with Miniflo-L™, which selects the serial data output type by GPS or Loran Manufacturer. 0 = none 1 = Allied Signal, KLN series (Bendix/King) 2 = AirData, used to communicate with a Shadin Airdata computer 3 = Arnnav, used to communicate with most Arnnav Loran or GPS 4 = Trimble, used to communicate with most Trimble Loran or GPS 5 = Generic, used to communicate with most Garmin GPS
I	X =	GPS/Loran Input: 0 = Off 1 = On
D(d)	x =	Endurance Warning Time: 0 = 45 minutes 1 = 5 minutes 2 = 10 minutes 3 = 20 minutes 4 = 30 minutes
F	x =	Filter Type: 0 = Injector 1 = Carburetor, for engines equipped with a carburetor
W(u)	x =	Ignore Loran Warnings 0 = No (default) setting used with Shadin Flow Meter. With GPS, set to zero (0). 1 = Ignore Loran Warnings. Used with Foster Loran only.
S	xxxx =	Low Fuel Level: Displayed in same units of measure as the flow rate.
* = Group 1 information		( ) = actual letter display. All others displayed as shown

## **7. DISPLAY BRIGHTNESS ADJUSTMENT**

The following paragraphs give the procedure to adjust the brightness of the display as desired.

1. Remove the Miniflo-L™ from the instrument panel. Do not disconnect the Miniflo-L™ from the aircraft harness.
2. Turn the aircraft master switch on.
3. With a small, flat blade screwdriver adjust potentiometer for the desired display brightness.



4. Turn the aircraft master switch off.
5. Re-install the Miniflo-L™ into the instrument panel.

## **8. SPECIFICATIONS**

Certification:	TSO-C44a
Maximum usable fuel:	1,800 gallons 6,822 liters 9,999 lbs 5,484 Kg @ 0.805 Kg/lit
Maximum Altitude:	40,000 ft
Operating temperature:	-30° to 50°C
Humidity:	Up to 95% @ 32°C
Accuracy:	± 2%
Ground Speed Range:	27-600 knots
Functions:	Fuel Flow (selectable endurance warning) Fuel Used Fuel Remaining Full Fuel Add Fuel Endurance NM/gallon Fuel to Destination Fuel Reserve

### **ELECTRICAL RATING**

Input Voltage:	14 – 28 VDC
Input Current:	200mA @ 14 VDC to 28 VDC

### **ELECTRICAL INTERFACE**

RS-232, Serial Data

### **MECHANICAL RATING**

Weight:	12 oz.
Dimensions:	3 ¼" x 1 ¼" x 7 ½"
Mounting:	Instrument Panel

**COMPATIBLE RECEIVERS:**

ARNAV	R15, R20, R21, R30, R40, R50, R50V, R50i, R5000, FMS5000, Star5000, FMS7000
BENDIX/KING	KLN-35, KLN-89B, KLN-90, KLN-90A, KLN-90B, KLN-94, KLX-135, KLN-900
BFGOODRICH/ FOSTER	F4, F14, 500, 501, 616, LNS-6000
GARMIN	100, 150, 150XL, 155, 155XL, 165, 250, 250XL, 300, 300XL GPS 400/500, GNC 420/520, GNS 430/530
MAGELLAN	5000
NORTHSTAR	M1 (above S/N 14800), M1A, M2, M3, 60, 600
SKYFORCE	(*AIM) KMD-150, Skymap II, Skymap IIIC
TRIMBLE	1000DC, 2000, 3000, 2100, 3100
IIMorrow/ UPS Aviation Technologies	Apollo 612B, 618, 618TCA, 820 GX-50/55/60/65, 360, CNX80 NMS 2001, NMC 2001

\*Avionics Installation Module

**PIN ASSIGNMENTS:**

PIN	Miniflo-L™ P/N 91204XT-D	TRANSDUCER 68050X	TRANSDUCER 6605XX
1	+14VDC or +28VDC (2A Circuit breaker)		
2	Airframe Ground		
3	Left/Single Fuel Flow Transducer Power (+12VDC to Transducer)	Red Wire (+12VDC input)	PIN A (+12VDC input)
4	Right Fuel Flow Transducer Power (+12VDC to Transducer)	Red Wire (+12VDC input)	PIN A (+12VDC input)
5	Ground to Left/Right Fuel Flow Transducer	Black Wire	PIN C
6	TX RS-232 (to LORAN or GPS)		
7	Left/Single Fuel Flow Transducer Signal	White Wire	PIN B
8	Right Fuel Flow Transducer Signal	White Wire	PIN B
9	RX RS-232 (from LORAN or GPS)		

## **9. WARRANTY INFORMATION**

### **Limited Warranty**

a. Warranty Period. For all Products purchased by Buyer, the Warranty Period shall be twelve (12) months from the date of the installation by Buyer into its end-customer's product or twenty- four (24) months from the date of shipment to Buyer, whichever occurs first. Buyer will provide confirming installation date and applicable customer data to Seller upon request of Seller. The Product Warranty may be extended to end user customers of Buyer who purchase and receive such products within twenty-four months of the date the Products are shipped to Buyer.

b. During the Warranty Period, Seller warrants that:

1) All Products meet specifications published by Seller appropriate to the model and options purchased as of the shipping date or agreed in writing between Buyer and Seller; and

2) All Products will be free from defects in materials and workmanship or defects due to design (other than any design or requirements specified by Buyer)

c. For any failure covered by this limited warranty, Seller shall repair or replace, at Seller's option, the defective Product. Buyer will promptly notify Seller of any failure that occurs during Warranty Period and request an RMA from Seller and provide the reason for return in writing if Buyer returns the product for repair or replacement. Failure to provide such notice will relieve Seller of its warranty obligations.

d. Repair or replacement of a defective Product is the sole and only remedy under this warranty. In case of replacing a defective original Product, Seller will warrant the replacement Product for the remaining Warranty Period of the original Product

e. During the Warranty Period, replacement Products will be shipped on a no-charge basis on condition that all defective Products must be returned immediately upon receipt of replacement Products. Failure to return the defective Products promptly will result in an invoice for the full price of the replacement Product.

f. Notwithstanding the above, Seller provides no warranties (1) on any Products or portion thereof which is based on the Buyer's design; or (2) respecting any larger system of which the Seller's Product is only a component or part; and (3) non-Seller supplied replacement or repaired parts or Products.

g. This limited warranty shall not apply to any product that has been repaired or altered by any person other than Shadin Avionics or that has been subjected to misuse, accident, incorrect wiring, negligence, improper or unprofessional assembly or improper installation by any person.

h.. *This Warranty does not cover any reimbursement for any person's time for installation, removal, assembly or repair.* Shadin Avionics retains the right to determine the reason or cause for warranty repair or replacement.

i. This Warranty does not extend to any aircraft, vehicle, boat, machine or any other device to which this Shadin Avionics product may be installed, connected, attached, interconnected or used in conjunction with in any way.

j. Buyer is responsible for any damages caused by inappropriate transportation.

**k. THIS WARRANTY IS IN LIEU OF ALL OTHER WARRANTIES, EXPRESS OR IMPLIED, INCLUDING WITHOUT LIMITATION, WARRANTIES OF MERCHANTABILITY OR OF FITNESS FOR A PARTICULAR PURPOSE. BUYER'S EXCLUSIVE REMEDY FOR BREACH OF WARRANTY IS REPAIR OR REPLACEMENT, AT SELLER'S OPTION. ANY LAWSUIT BY BUYER AGAINST SELLER, ITS OFFICERS, DIRECTORS, EMPLOYEES, OR AFFILIATES, ARISING OUT OF OR RELATING TO WARRANTY CLAIMS REGARDING THE SUBJECT PRODUCT, SHALL BE FILED WITHIN TWO YEARS FROM SHIPMENT OF THE SUBJECT PRODUCT TO BUYER BY SELLER.**



**Digital Fuel Management System Data (Miniflo-L™)**

Part Number: \_\_\_\_\_

Serial Number: \_\_\_\_\_

Left/Single Transducer Part or Kit Number: \_\_\_\_\_

Left/Single Transducer Serial Number: \_\_\_\_\_

Right Transducer Part or Kit Number: \_\_\_\_\_

Right Transducer Serial Number: \_\_\_\_\_

Installation Date: \_\_\_\_\_

Installed By: \_\_\_\_\_

**Group 2 Configuration Selections**

<b>O</b>	Serial Output Type	
<b>I</b>	Serial Input On/Off	
<b>D(d)</b>	Endurance Warning Time	
<b>F</b>	Filter Type	
<b>W(u)</b>	Ignore Loran Warnings	
<b>S</b>	Low Fuel Level	

NOTES:







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